

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SA4025WE

This certificate, issued to Hamilton Aviation Company
P.O. Box 11746
Tucson, AZ 85734

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 4b of the Civil Air Regulations.

Original Product — Type Certificate Number A-793
Make General Dynamics
Model 240-27, 240-52

Description of Type Design Change: Installation of cargo door, cargo floor, class "E" cargo compartment and provisions for weather radar installation in accordance with FAA sealed Hamilton Aviation Company Master Drawing List 3614500, Revision K or later FAA approved revision.

Limitations and Conditions: The approval of this change in type design applies basically to the General Dynamics Models 240-27 and 240-52 only. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any of those other previously approved modifications will

See Continuation Sheet 3

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: April 4, 1979

Date issued:

Date of issuance: September 28, 1979

Date amended: Oct. 20, 1981



By directing of the Administrator

J. Kasnowski
(Signature)
Chief, Engineering & Manufacturing Branch
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

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(Continuation Sheet)

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'28/79 (I)
10/20/81 (II)

introduce no adverse effect upon the airworthiness of that aircraft. This determination should include consideration of significant changes in weight distribution such as an increase in the fixed disposable weight in the fuselage. This approval does not include a cargo restraint system. This modification may only be incorporated on aircraft which are operated unpressurized.

- END -

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FAA FORM 8110-2-1 (10-69)

This certificate may be transferred in accordance with FAR 21.47.

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